

JUNE 13, 1921

Issued Weekly

PRICE 15 CENTS

AVIATION AND AIRCRAFT JOURNAL



Aircraft versus Battleship

Illustration by F. W. Wilson

VOLUME X

Number 24

SPECIAL FEATURES

THE MORGANTOWN CRASH

FRENCH AVIATION IN 1921

"WHO'S WHO IN AMERICAN AERONAUTICS"

AIRPLANE PERFORMANCE AND DESIGN CHARTS

DIRECTORY OF AIR SERVICE ACTIVITIES

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AVIATION AND AIRCRAFT JOURNAL

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Vol. X June 15, 1933 No. 24

CONTENTS

Aeronautical Legislation

ALTHOUGH the United States is still awaiting the coming of national legislation for the regulation of air navigation which would bring order into what is now chaos, it must be said that bills pending to rapidly state a state of affairs are wanting. In fact, there is right now a mass of proposed aeronautical legislation before Congress. The reason bills that have been introduced at this session often develop on one another's heels is such extent that the result may be extremely confusing to our lawmakers.

Not only have bills been presented which propose the creation of separate departments, but amendments, hearings and reports have been proposed, all of which none or less answer the same purpose. All kinds of regulatory measures have been introduced as well as many special bills calling for all sorts of appropriations from the purchase of aircraft patents to the payment of losses of the NC-4.

It would really require a lawyer's full time to read and to consider carefully all these measures. What will come out of this session is problematical. For the time being the various bills are in several committees. Some important "orders" have also been introduced at appropriation bills. All this hangs on such a flimsy foundation as to be handled successfully by having Aviation Committees in both Houses and the sooner they are authorized, the quicker will there be an intelligent consideration of aeronautical laws.

The Lesson of Morgantown

THE Morgantown crash, the worst that has ever occurred in the history of American flying, should serve as an urgent warning to our lawmakers that no time should be wasted in the making of such legislation as aviation requires for the safety of those engaged in its pursuit.

It is hardly a repetition of what we have so often expressed in these columns to say that our regulations in order to achieve safety would chiefly be of a well planned set of airways and a comprehensive system of weather forecasting lack of suitable landing fields often forces aviators to jeopardize their safety by coming down on most unsuitable sites. Lack of reliable weather forecasting keeps them in ignorance of the atmospheric conditions they will meet en route. The two factors combined make a poor insurance for safe flying.

The Morgantown disaster was probably due to both of the above named causes. While it is possible that an emergency landing field may not have saved the Eagle from crashing, an accurate weather forecast, showing Lowley Field in town, would have prevented the take-off of the ill-fated airplane. The fact that all the machines which took off at that time flew straight into a terrific pile, from the effects of which even an extremely maneuverable airplane was wrecked, would seem to show that the weather was chiefly responsible for this disastrous accident.

It is therefore gratifying to learn that the United States Navy, in cooperation with Weather Bureau, has decided to issue daily weather forecast balloons which will be available to all who fly. This measure is however only a modest beginning. To begin with, the balloons, which since June 1 is being sent out from the radio station at Arlington, cover only the eastern half of the United States and Canada. Second, it will be available for private use only on airplanes equipped with a radio station—and these are extremely few in number. Third, an aviator may start several hours before the bulletin is issued (at 20 a. m.) and suffer loss a few days before the ground stations over which he may fly are even in line. This would seemingly make it desirable for our ports to fly alone signals such as are necessary in harbors. Putting aside all these little bits of the warning in time and further, without remarking on the very noticeable effects of the government's present emergency this service, it seems awkward that it should require three separate government departments—the Navy, the Weather Bureau, and the Signal Corps of the Army—to issue weather forecasts to them.

Regardless of these minor points, however, the issuing of daily weather forecasts is bound to prove of the greatest benefit to them, one of which they should take the fullest advantage, for it is another step toward greater safety.

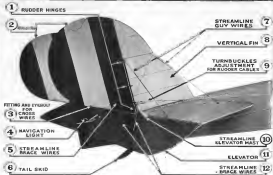
No More Civilian Stunting

AMERICAN civilian aviation, which has been conspicuously free from serious accidents ever since the termination of legislation prohibited the resumption of passenger activities, recently experienced a regrettable series of fatal accidents. During very few instances, the real cause of these fatalities was stunting in all of its forms.

It is practically impossible to draw a line between what is safe and what is unsafe in stunting. Consequently, the only sensible measure that will prevent a repetition of such accidents is the absolute prohibition of civilian stunting, be it over public airfields or elsewhere.

Passant pilots have to be trained in stunting because aerial fighting requires it; but commercial pilots need not and should not start transport airplanes, for the temptations are too strong to do it safely. Nor are masters of aerobatics of aerobically the most desirable commercial pilots. Taking from European experience, the ordinary owner rather than the pilot. The British and French air lines employ former bombardment pilots in preference to private pilots because the former's staying power is of greater importance for commercial aviation than brief flashes of brilliancy.

What American commercial aviation needs is order to give the confidence of the public in safe and sane flying; not stunting. Hence stunting of civilian aircraft should be prohibited, as it is in Canada and Great Britain.



The Empennage

As a chain is said to be no stronger than its weakest link, so an airplane can be no stronger than its control surfaces.

The quality of its engineering and the sincerity behind its craftsmanship are no-where better typified than in the empennage or tail group of a Glenn L. Martin airplane.

The pilot constantly faces a problem infinitely more difficult than that of the railway engineer or the ship's navigator. With an equipment weighing tons, he bestrides the empty air. Aided only by instinct and the instruments in the cockpit, he must check his equilibrium and pursue his plotted course. The elevator and rudders must be under instant control to guide him safely in his flight.

That The Glenn L. Martin Company, through years of progressive engineering, has won an enviable reputation for its planes is in part due to the ease of controllability and trustworthiness of construction which has always been a feature of the empennage or tail group.

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The Morgantown Crash

AVIATION AND AIRMAIL INTEREST deeply agree to have to coincide by far the worst accident that has yet befallen American aviation. On Saturday, May 28, a Curtiss-Eagle ambulance airplane of the Army Air Service which was flying from Langley Field, Hampton, Va., to Bolling Field, Washington, D. C., was caught in a violent storm and was wrecked and crashed near Morgantown, Md. The seven occupants of the machine, all of whom were killed, were:

Col. Arthur Miller, Air Service, General Staff College, instructor of the Congressional School of Aeronautics.

Major General Murray Connelley, representative of the Curtiss Aeroplane & Motor Corp. in Washington, D. C., a Major in the Air Service during the war and formerly a member of Congress from Iowa.

A. G. Batschelder, chairman of the executive board of the American Aeronautical Association and member of the board of governors of the Aero Club of America.

Col. Lester Hunsinger, Air Service, of Bolling Field, pilot of the Curtiss-Eagle.

Second Lieut. Cleveland W. McDermott, Air Service, of Langley Field.

Second Lieut. John M. Pennewill, Air Service, of Langley Field.

Sergeant Richard Hunsenbaker, Air Service, of Bolling Field.

On Another Flight

The big ambulance airplane and four smaller craft left Bolling Field early Saturday morning for a round trip practice flight from Washington to Langley Field. These flights the airplane and the guests aboard the Curtiss-Eagle made the trip also to witness the review of the fully equipped air brigade which is to take place in the army and navy landing tents off the Virginia Capes late in June. Langley Field is 150 miles in an air line south of Washington.

The Curtiss-Eagle was under the command of Lieutenant Ames. In addition to Mr. Connelley, Mr. Batschelder, Mr. Connelley and Capt. Hunsinger were aboard as guests. Representative Philip P. Campbell of Kansas, Representative Joseph Walsh of Massachusetts and Captain Guy de Lersperre, Air Attache of the French Embassy at Washington. Representatives Carroll and Walsh and Captain de Lersperre made the return trip to Washington by night steamer from Norfolk.

At Langley Field two other passengers, Lieutenants McDermott and Pennewill, both of whom were formerly stationed at Kelly Field, Texas, were taken on board for the return trip to Washington.

The big airplane had been stripped of its ambulance as accommodations for this trip and seats had been provided for the passengers.

The distance of 180 miles above the valley of the Potomac past Mount Vernon, Quantico Hall and other historical Colonial homes, was made without mishap by the ambulance airplane.

Run Into Storm

The airplane had covered 120 miles of the return trip before running into a storm which nearly played havoc with fear other army airplanes, including one in which Brig. Gen. William Mitchell, Assistant Chief of Air Service, was engaged in a round trip flight, between Bolling and Langley Fields. The returning planes were in the vicinity of Indian Head, Md., when one of the seven returning planes a forced landing was made when they ran into the storm.

Captain Richard S. Wright, an old friend of Gen. Mitchell, who was flying a Fokker airplane from Langley Field to Bolling Field, was caught in the same storm and forced a landing at a small farm, where his plane was damaged.

Brig. Gen. Mitchell, flying in a Fokker airplane, and Captain Wright, flying in a Fokker airplane, were caught in the same storm and forced a landing at a small farm, where their planes were damaged.

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Sam C. Ocker of Bolling Field was caught in the edge of the storm, but managed, after great difficulty, to ride out of the peril and return safely to Bolling Field half an hour after the accident to the ambulance airplane occurred.

An airplane commanded by Captain Turner of the West Coast Air Service was caught in the storm and forced to land at Quantico, being unable to make its way through the Washington.

What happened inside the Curtiss-Eagle when the airplane struck the storm over the Potomac may never be known. The time of the crash was 6:25 P. M. as indicated by the fact that Mr. Connelley's watch, when found, was stopped at the moment. While the accident was witnessed from a distance both by persons stationed at the naval proving ground at Indian Head, on the Potomac, and on board the naval yacht Delphin, it is understood that only one person, a radio operator, reached the scene of the accident in the early hours of the evening.

Those at Indian Head who saw the plane say that the pilot appeared to endeavor to bring the airplane to earth in

the storm in the full fury of the storm. Lieutenant Ames was seen to strike the water three times evidently looking for an open space to land. Then he found his landing was suddenly caught by a sudden updraft just which caused the airplane to flip perpendicular.

General Mitchell said that he watched as plainly whatever to look. Ames, the pilot of the big plane, and reported that the plane in the water could have been able to ride and the storm in which the airplane was caught.

Press Work of Last, Ames

"Lieut. Ames was one of the ablest of the younger pilots in the service," said Gen. Mitchell. "He was a former officer and was pilot at Bolling field, and had taken this plane into the air before. He made a splendid fight with the big Eagle yesterday morning from Bolling field to Langley field. He has been serving in that post at Bolling field for about a month, and comes from Boston. During the war he entered the air service from civil life and got to England during the closing months of the war. He is one of the men taken into the air service during the war who have been commissioned in the regular army under the provisions of the army reorganization act of June 4, 1920."

Colonel Miller's Career

Colonel Miller was born in Chicago on Sept. 23, 1878. He attended the public schools of the city. He was an A. B. man of St. Mary's College, Kansas, and St. Louis University. At the outbreak of the Spanish-American War he enlisted with the State volunteers and saw service in Cuba, reaching the rank of First Lieutenant. In 1901 he went to the Philippines for a year, being serving later in the regular army cavalry. The action which won for Colonel Miller the Congressional Medal took place on this island on July 2, 1899. The Americans were opposing hostile Moros and Colonel Miller, under heavy fire, with the assistance of enlisted men, placed a machine gun in advance of the former position, at a distance of about twenty yards from the enemy.

When the United States entered the World War, Colonel Miller was assigned to the Air Service with the rank of Lieutenant Colonel and was sent to Kelly Field as commanding officer. He subsequently was commanding officer at the aviation camp at Waco, Texas, and Camp Greener, North Carolina, in which he presided with the rank of major at the several training camps established on Long Island.

After the armistice Colonel Miller was appointed as assistant to the Director of Military Aeronautics. Later he was in command of Air Service activities on Long Island. In February, 1920, he returned to his old post of commanding officer at Kelly Field, Texas, for a time. At the time of his death he had just completed a course at the General Staff College in Washington, in which he graduated with high honors. He is survived by a widow and several children.

Connelley's War Service

Murray Connelley was born in Dubuque, Iowa, and at one time represented his district in Congress. During the war he was a Major in the Air Service, being assigned at various times as executive officer at Wilkes Wright Field, Fairbury, Neb., commanding officer at Headquarters Field here and as the Chief of Staff of the Air Service. He was graduated from Cornell University and took post graduate courses

at Oxford College England, and the University of Heidelberg. Connelley was born in 1877. He was married.

Mr. Connelley was a representative member of the Executive Committee of the Smithsonian Institution, and had many important connections in Iowa. He was a delegate-at-large to the 1918 convention of the Democratic Party. His claim was the Aero Club of America, the National Press Club and the Metropolitan Club.

Mr. Batschelder was a former newspaper man and was a native of Utica, N. Y., where he was born forty-eight years ago. He was at one time editor of Motor, and it was while serving in such that he represented the American Aeronautical Association, which gave under his leadership to an organization of large size and influence. Mr. Batschelder had made his home in Washington since the Aeronautical Association spread national headquarters there.

Lieut. Ames Once Held by Mexico

Lieutenant Stanley Ames had been an aviator for several years. Enlisting in the United States Army at the outbreak of the war, he was transferred to the American forces in France. In June of last year he made a forced landing on Mexican territory and was held by the Mexican authorities at Matamoros. His release was ordered by General P. Elias Calles, Mexican War Minister. Lieutenant Ames was testing officer at Bolling Field, just outside of Washington, and was regarded as an authority on auto engineering and testing.

Lieutenants McDermott and Pennewill were stationed at Langley Field, but they were recently transferred from Kelly Field, Texas, for the forthcoming joint army and navy biplane trials.

MURDER CONSPIRACY

John, Lieutenant McDermott, who came from Syracuse, taught in France and was officially credited with having downed three German planes. The French awarded to him the Croix de Guerre and the Médaille Militaire. He likewise received the Distinguished Service Medal.

Speaking for the membership of the Aero Club of America, Maurice Clancy, managing director, said:

"The loss of these men, all of whom were members of the staff of a national eye, Colonel Miller, Mr. Connelley and Mr. Batschelder were leaders in aeronautical development in this country and their places will be hard to fill. Enduring as much as aviation that they were willing to fly, they were indeed martyrs to a great cause—the development of the airplane as a mode of transport."

Investigation Underway

As soon as the accident became known in Washington the Chief of Air Service ordered a thorough investigation into the cause of the accident. The findings of the special board appointed for this purpose have not yet been made public.

Statement by the Curtiss Co.

The following statement relative to the Morgantown crash was issued by C. M. Keys, president of the Curtiss Aeroplane & Motor Corp.

"I have received the proper orders of the Government that we should do a full investigation of the crash of the Curtiss-Eagle and that we hope the findings of this investigation will be made public."

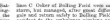
"We have complete faith in the ship, and are willing to rest upon the competency of the Army Air Service and declare that we have flown it during more than a year of safe and successful operation."



Gen. ARTHUR MILLER, Air Service, U. S. A.



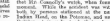
Capt. RICHARD HUNSENBAKER, Air Service.



Capt. TURNER of the West Coast Air Service.



Capt. HUNSENGER of the Air Service.



Capt. HUNSENGER of the Air Service.



Capt. HUNSENGER of the Air Service.



Lt. AMES, who was once held by Mexico.

Airplane Performance and Design Charts

By L. V. Kachner

Aeronautical Engineer, Engineering Division, Air Service
(174994), 1921, by G. V. Kachner

The subject matter of the present report is a graphical solution of an empirical-theoretical method of predicting the performance of an airplane, of which are known the weight, size, horsepower and external characteristics, and, by reverse, a method for determining the weight and size of an airplane of given horsepower and external characteristics, which it is to be designed to make a particular performance.

The example which serves as an illustration of the method, in order to be of practical value in this country, is connected with the "Standard Atmosphere" adopted by the Engineering Division and with the slope of the lift-drag curve. The curves characteristic of airplanes equipped with various airfoils. The method may be applied to the construction of charts consistent with any particular standard of atmosphere, such as the British and applicable to airplanes equipped with any new type of airfoil, such as the Hinesley-Pearce wing or the British "Auk" wing. It will be shown that the ultimate functioning of a supercharged engine can be achieved by the present method.

The main variables entering into any consideration of airplane performance are the loading per horsepower, the loading per square foot of supporting surface, and one which will be called "fineness," and which will be a measure of the (three-fourth) efficiency or lift/drag ratio of the complete airplane. For instance, if the "fineness" of one airplane is 100 and that of another is 120, it means that at any value of K_y common to both, the L/D of the latter is (120/100) or (1.2) times greater than the L/D of the former. Since the horsepower required is directly proportional to drag, and hence inversely to L/D , it means that at a given value of speed or of K_y and W , the second airplane can have loading per horsepower (1.2) times as great as the first airplane. The "fineness" which appears throughout this report is then always the ratio of the L/D of an airplane relative to the L/D of a certain basic airplane with a "fineness" of 100. The value of this ratio has been taken in order to confine the values to convenient limits, make scale interpolation, and the scale has been multiplied by 100 so as to deal in round numbers.

Second, purely empirical performance charts have been presented in the past, and they are practically worthless because of the omission of "fineness." For instance, these charts would give for the same loadings per horsepower and per sq. ft. identical performance for a large four-engine bomber as for a small single-engine fighter airplane, whereas it can be shown that the L/D ratio of the latter may be as much as 160 per cent greater than that of the former. Differences in load speed at the ground would in this case be of the order of 55 per cent. A further disadvantage is that these charts are not extended beyond the loadings in present practice. Purely theoretical analysis of performance have been presented, and satisfactory results have been secured for long and tedious mathematical computations. The construction of a recent work of this nature emphasizes the need of some complete empirical data on (1) wings, to determine the effects of various camber lines, (2) parasite drag, to determine the resistance coefficient of the bodies, (3) the effect of interference, (4) propellers, to determine the thrust and torque, (5) engines, to determine the drag-of as horsepower per square foot of airframe.

Obviously the logical solution of this problem is by an empirical-theoretical method. The methodology of the proposed method may be overcome by taking into consideration "fineness." The difficulties of the theoretical methods may be obtained merely by solving the effects of (1), (2) and (3) above, by including them all under a factor "fineness." The explanation of the preparation of the charts will indicate in what manner the performance of the engine-propeller will be obtained, and it will be shown that to what extent the theoretical questions of flight have been incorporated.

The method of this report consists, (a), in the construction of an empirical-theoretical chart for the determination of "fineness," means-percentage performance, high speed at any altitude, and in the best of the approximate, absolute velocity; (b), in the construction of an empirical-theoretical chart for the determination of rate of climb at the ground; (c), in the use of an equation for time of climb to any altitude based on the theory that rate of climb is a direct function of the altitude; and (d), in the construction of the results of (a), (b) and (c) to construct, with the aid of cross-plotting, a performance and design chart. This final chart furnishes a complete picture of the variation of all of the important ground level, cruise ceiling and absolute ceiling, of horsepower, high speed, time of climb and rate of climb, of an airplane with a given engine and with any combination of the three variables—loading per horsepower and per sq. ft. and "fineness."

Speed-Altitude Chart

The fundamental chart which serves as a means for studying the "fineness" of airplanes and the functioning of the engine-propeller at an reduced density, is a speed-altitude chart

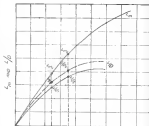


FIG. 2

which is based on the definition of maximum horizontal velocity attainable at any altitude as the velocity at which the horsepower required by the airplane is consumed at its maximum L/D equal to the horsepower available from the engine-propeller. In constructing such a chart it is thus necessary to study the variation of these features with altitude.

At zero altitude the horsepower required for a certain basic airplane is known from a wind tunnel model test. For any other airplane B will be expressed in terms of the "fineness" as relation to the basic airplane. Horsepower available at zero altitude is known from dynamometer test and by assuming a propeller efficiency. At altitude, horsepower available drops off in a manner which is determined experimentally. At altitude, horsepower required and also velocity depend on density, and these two variables are studied theoretically.

The presentation of all these considerations in graphical and tabular form of a series of continuous to the original loading

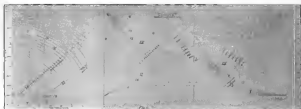


FIG. 3

per horsepower scale, which is used indiscriminately to represent both H/D required and H/D available, since for any particular solution they are equal. It will be understood that for any solution for velocity, the H/D of the airplane under consideration is connected for its "fineness" and at altitude of consideration, so that it is changed into an equivalent H/D of the basic airplane with "fineness" 100 and flying at zero altitude. In addition the velocity so found is corrected to be consistent with the density at the altitude at which the airplane speed consideration is desired.

The H/D is taken as the basic airplane in the example of this report. According to our definition of "fineness"

$$(1) \text{ "Fineness"/100} = (L/D)_{\text{airplane}} / (L/D)_{\text{basic}}$$

where $(L/D)_{\text{airplane}}$ and $(L/D)_{\text{basic}}$ are compared to the same value of K_y and W . For B compared to the same value of K_y and W then at any value of lift or K_y common to both airplanes, the drag and hence the horsepower required is proportional to (fineness/100). The horsepower required, HP_{req} , if reduced to the HP of the basic airplane becomes $HP_{\text{req}} / (F/100)$ and the loading per horsepower becomes $(W/HP) / (F/100)$. In other words, if the "fineness" of an airplane is greater than that of the H/D basic, then the HP of the H/D will be greater and the H/D will be less, in order to attain any particular speed.

The following fundamental equations are of use in finding the desired relations:

$$(1) \text{ } HP = W / (55 \times F / (100 \times V))$$

$$(2) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(3) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(4) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(5) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(6) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(7) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(8) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(9) \text{ } H/D = W / (55 \times F / (100 \times V))$$

$$(10) \text{ } H/D = W / (55 \times F / (100 \times V))$$

The speed V found from (7) is used in (8) and the L/D of (4) and the L/D of (10) correspond to the same angle of incidence in Fig. 2. Equations (5) and (6) are solved for various angles of incidence from zero to 10° to determine L/D , and again for various values of the total weight of the airplane, the area being considered not to change. The relation between H/D , K_y , H/D , and F is then obtained and has been plotted in Fig. 2, curve 1, where "a" is the speed scale, "b" is the H/D scale and H/D is the parameter.

Next will be considered the effect of altitude on the disposition of curves 1. At any constant angle of attack and hence constant value of K_y , the speed at altitude compared with the speed at zero altitude will be

$$(1) \text{ } V_a = (1.3)^{H/1000} V_0$$

where V_0 is the relative density at the altitude and equals $H/1000$. The HP at this altitude and speed will be

$$(1) \text{ } HP_a = (1.3)^{H/1000} HP_0$$

scale "b" of H/D as now placed along scale "a" in such a manner that if we start from "a" and go vertically to the zero altitude line of curves II and then horizontally to "b", we would find there the curve H/D as at "b". Starting again from a point on "a", going vertically to the 6000 ft. curve, then across to "b", we will have multiplied the H/D of "a" by $(1.3)^{6000/1000} = 1.675$. Flying at a given speed at some altitude other than zero altitude is equivalent to flying with a heavier weight at zero altitude. The H/D as a term of the basic airplane will be greater by $(1.3)^{H/1000}$ and curves II effect this multiplication by $(1.3)^{H/1000}$. High speed is then read on scale "b" instead of on scale "a".

A further correction to H/D , as used applied and takes into account the effect of "fineness." As indicated above, the H/D scale must be modified by multiplication with $1/(F/100)$. This is effected by curves IV in starting from scale "b" to "c". Starting from a H/D of any W on a scale "b", moving to the right to a function of any W on curve IV, a vertical step to "c" would indicate a H/D corrected for "fineness" of $15/(110) \times 5.8$. Moving now vertically from curve IV to "a", say, the 5000 ft. line of curve II, we would find on "a" a H/D of $7.5 \times 1.035 = 7.76$. Moving now to the proper H/D on curve I, the curve of L/D , then vertically to the 6000 ft. line of curve III, then to scale "b", we have the resultant high speed.

H/D has not yet been corrected for "fineness" and for altitude has been modified for altitude. There remains only to correct H/D available due to the variation with altitude of the engine-propeller performance. Here "a" is first moved over horizontally to scale "b". Scale "b" is an altitude scale and scale "c" is an engine-propeller scale. Suppose we have a specific engine curve VI. Starting on "b" at zero 2000 ft., moving to the right to curve VI we would find on "c" a factor less than unity which would represent the horsepower available from the engine-propeller at 2000 ft. in terms of the horsepower available from the engine at zero altitude and with a propeller of 100 per cent efficiency. This means that to attain a given speed at 2000 ft., the H/D available at zero altitude would have to be increased by $1/66$ if the curve VI are to be used. The curve V of H/D are as constructed as to effect this multiplication. So that now if we came from 2000 ft. on "b" to the proper engine curve VI, then to the H/D at zero altitude on curve V, then to the "fineness" on curve IV, then to the 6000 ft. line of curve II, then to the proper H/D on "b", then to scale "a", then to the 5000 ft. line of curve III, then to scale "b", we will find there the high speed which it is possible to attain at 5000 ft. with the "fineness" and the loadings per horsepower and per sq. ft. of the airplane investigated. The same procedure applies to all other altitudes.

Landing Field Notes

Milford, Iowa

Milford, Iowa, has a permanent airfield one-half mile N of Milford and one-half mile S of Lake Okoboji. This field which is operated by Donaldson Bros. Aero Club, opened for Curtiss airplanes and exhibition aerobatics, was opened in January 1933. Visiting air pilots welcome. There is a large hangar on the field which accommodates six airplanes and facilities are available for all kinds of repair and overhaul work.

Hartford, Conn.

The first municipal landing field in any Eastern City has now been completed at Hartford, Conn., and has already been visited by a number of them. The city has so far expended more than \$10,000 to put the field in condition and, made from proving supplies of gasoline and oil, the field also offers to have the accommodation of a club house. An attendance is kept constantly on guard. The Hartford municipal aviation commission wishes it understood that there from any place, at any time, are welcome to "drop in" on the field. Recently several army fliers from Pennsylvania, Mass., visited the Hartford field and purchased 1 one of the field they had flown on. There is a total space of 100 acres and the aviation has a straight road, one service road, 1,000 ft. in one direction and 1,000 in the other. Markers are now being installed and, aside from the usual symbols for the information of the flier, the word HARTFORD will be seen out in white concrete. The customary signal direction, size and other safety appliances are provided.

Horton Perry Mason, inventor of the Mason elevator, is chairman of the municipal aviation commission at Hartford, and is also president of the Aero Club of Hartford, established in 1929. Aviation visitors Hartford should phone Mr. Mason or one of the other aerial commissioners. They will find a cordial welcome. Mayor Bennett of Hartford and other municipal officials are decidedly friendly toward flying and are cordially greeting the aviation commission.

Fliers intending to visit Hartford whenever possible should send a line in advance of their arrival addressed to "Aviation Editor - Hartford Courant," and that newspaper, the largest morning paper between New York and Boston, will see that their coming is heralded.

Stamaton, Va.

Stamaton, Va., has put itself on the aerial map by establishing a municipal landing field in accordance with U. S. Army Air Service specifications. E. W. Pope of the aviation landing field committee of the Chamber of Commerce has sent in the questionnaire check upon the following situation: Name of field, Lake, shape and dimensions in feet, 600 x 400, direction of long axis, North-South; Markers, according to A. S. specifications; Contour of field, level on west side, slight rise to north, with level roadway in center; landing possible in west, northeast, north, east and south; no low fence around entire field, telephone wires at north end, ditch on east side, but all at such distance from runway as to be negligible, according to pilots who have used field. Small aerobatics would be possible, high grade of vegetation available at Beverly Garage, 1/2 mile from field in business section. Field located on northern outskirts of town, about one mile north-west of intersection of Chesapeake and Ohio, and Valley Branch of Baltimore and Ohio Railroads. Altitude above sea-level, 1,000 ft. Stamaton and Augusta Chamber of Commerce is operating field, Clayton D. Babson, Secretary, official interested. Field open to all pilots, Lynchburg Air Service Corp., Lynchburg, Va., now using this field.

Rochester, N. Y.

Authorization of a vote issue of \$125,000 to provide funds to pay for the Southern road aviation field at Rochester, N. Y. and the municipal wharf at Rochester was made recently by the Rochester Common Council. Fifty thousand dollars will cover the aviation field while the remainder of the money will be used for the wharf project.

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